## US Department of Transportation Federal Aviation Administration

## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

Office Identification

50850017

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

for each such violation (Section 901 Federal Aviation Act of 1958).								
1. Aircraft	Make REPUBLI	C MA			_			
	Serial No. 946			Nationali	ty and Registration	Mark		
	Name (As shown on registration certificate)			Address (As shown on registration certificate)				
0 0	RUW HIRE THUR! WHE PICER PRINT WAR			i Rich Rich Bio King F. Comments				
2. Owner	HERE BOTH SING SING SIGNATURE DESIGNATION DE PROPERTIES DE PROPERTIES DE SONO EN TOUR PARTIES DE PROPERTIES DE SONO EN TOUR DE PROPERTIES DE PROPERTIES DE SONO EN TOUR DE PROPERTIES DE PROPERTIES DE SONO EN TOUR DE PROPERTIES							
This every of Doronium is 132 For FAA Use Only 2013 Square manife doing do do do do do do 132 For FAA Use Only 2013 Square manife do do do do do do 132 For FAA Use Only 2013 Square manife do								
io contour as was previous.								
Skin was reattached in accordance with AC43.13-1A. Chapterl. Section 3.								
this beneformanism some kniedle bil music pack of docused								
			4. Unit Identification		1		5Type	
Unit			Model		Serial No.		Repair	Alteration
	mi huggvit ha	ेर बर्धा हैंदे हैंदे	SHAMI	this won density		AUGE/		
AIRFRAME		ibed in Item 1 above)						
				· · · · · · · · · · · · · · · · · · ·				<del></del> -
POWERPLANT								
भूति के								
PROPELLER			s bioidsbuin	SHEET	o beliamen			
	Type			272 (2	Krighter Ester E	1. 15 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1.	23778564.55	
	Type	Process and the second	mental and the second	* ** All the State **	tillane (i the state of			
APPLIANCE	Manufacturer							
6. Conformity Statement								
A. Agency's Name and Address  B. Kind of Agency  C. Certificate No.								
HENRY RUZAKOWSKI POBOX 497			U.S. Certificated Mechanic			AP 267490854		
			Foreign Certificated Mechanic		nic			
			Certificated Repair Station					
TAV	ERN188 Fc. 3:	Manufacturer						
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.								
Date Signature of Authorized Individual								
API	211 4, 1998	5/						
7. Approval for Return To Service								
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED   REJECTED								
	A Fit. Standards				Other (Specify)	<del></del>		
I I .	pector	turer	Inspection Authoriza	ation				
<u></u>	A Designee Repair St	ation	Person Approved by Canada Airworthine					
Date of Approval or Rejection Certificate or Signature of Authorized Individual								
4/4/98 Designation No.  AP267490854IA S/////,								
			<del></del>			<del></del>	<del></del>	

## NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

## BOW DOOR REPAIR

Removed damaged bow door and installed serviceable unit. Outer skin was not on serviceable unit and lower 2/3 of door outer skin was used from damaged unit. Minor imperfections were hand hammered to return this part to contour as was previous.

Skin was reattached in accordance with AC43.13-1A, Chapter2, Section 3, Paragraph 99-c, d and Figure 2.18. Original spot welds were replaced with AN426AD3 rivets.

Manufactured new skin for forward section of bow door and riveted in accordance with AC43.131A, Chapter2, Section 3, Paragraph 99 d and Figure 2.18.

Bow door hinges were reattached and functional check for operation was satisfactory. Reinstalled original windshield and rubber molding. Weight change was negligible.

☐ Additional Sheets Are Attached